## (U) What is claimed is:

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1	(C) 1. In a semiactive radar guidance system for a
2	guided missile, such system including a heterodyne
3	receiver wherein the frequency of a first local
4	oscillator is required to be rendered coherent with
5	the frequency of echo signals from a target being
6	tracked by a frequency control signal from automatic
7	frequency control circuitry, such circuitry being
8	susceptible to vibration during flight to assume
9	one of two stable conditions, the first of such
10	conditions being one in which the frequency control
11	signal has the proper amplitude and polarity to
12	maintain coherency and the second of such conditions
13	being one in which the polarity of the frequency
14	control signal has an incorrect polarity, the
15	improvement comprising:
16	(a) first means, responsive to vibration-induced
17	changes in the automatic frequency control
18	circuitry from one stable condition to the
19	other, for generating a control signal
20	indicative of such change; and
21	(b) second means, responsive to the control
22	signal, for correcting the polarity of the

frequency control signal.

1	(E) 2. The improvement as in claim 1 wherein the
2	first means is a differentiator responsive to change
3	in the polarity of the frequency control signal.

1 The improvement as in claim 2 wherein the second means comprises a phase lock loop incorporating 2 the combination of a synchronous detector having a 3 first and a second input and an output terminal, a voltage controlled oscillator and a narrow band 5 6 summing amplifier, a signal representative of a target being tracked being applied to the first input 7 terminal, the output signal of the voltage controlled 8 oscillator being connected to the second input 9 10. terminal, the summing amplifier being disposed in circuit between the output terminal and the voltage 11 controlled oscillator with the control signal applied 12 13 to a second input terminal of such amplifier.

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